

METHOD TO ELIMINATE UNDULATIONS IN A COMPOSITE PANEL

ABSTRACT OF THE DISCLOSURE

5 A method and article of manufacture of joining a stringer to a wing or tail panel and curing a wing or tail panel is disclosed. A sheet of titanium is placed between a cured stringer and an uncured wing or tail panel. The stringer, wing or tail panel and sheet are then placed within a vacuum bag, and gas is pumped out of the bag. The bag is then placed in an auto-clave to cure the wing or tail
10 panel.